

A novel numerical approach for fracture analysis in orthotropic media

Nguyen Ngoc Minh, Nguyen Thanh Nha, Bui Quoc Tinh, Truong Tich Thien

Abstract — This paper presents a novel approach for fracture analysis in two-dimensional orthotropic domain. The proposed method is based on consecutive-interpolation procedure (CIP) and enrichment functions. The CIP were recently introduced as an improvement for standard Finite Element method, such that higher-accurate and higher-continuous solution can be obtained without smoothing operation and without increasing the number of degrees of freedom. To avoid re-meshing, the enrichment functions are employed to mathematically describe the jump in displacement fields and the singularity of stress near crack tip.

The accuracy of the method for analysis of cracked body made of orthotropic materials is studied. For that purpose, various examples with different geometries and boundary conditions are considered. The results of stress intensity factors, a key quantity in fracture analysis, are validated by comparing with analytical solutions and numerical solutions available in literatures.

Index Terms — consecutive-interpolation procedure, crack analysis, enrichment functions, orthotropic materials, stress intensity factor

1 INTRODUCTION

Thanks to its specific high strength and stiffness per unit weight, orthotropic

composite materials have been widely used in many modern engineering applications such as automobile industries, shipbuilding and aerospace components. Due to the demand to improve the durability of those structures, studies on fracture behavior of orthotropic media has arisen as an important and indispensable task.

Analytical investigation on fracture mechanics of orthotropic materials have been presented for some particular problems with relatively simple geometries and boundary conditions [1, 2, 3]. For more complicated problems, as usually encountered in engineering structures, numerical approach is more suitable.

Currently, the finite element method is the most popular which is widely used in both academic and industrial communities due to its simplicity and efficiency. To avoid the cumbersome task of re-meshing in modelling cracks, the extended finite element method (XFEM) was proposed [4]. In XFEM, cracks are not directly modelled as geometric discontinuities. Instead, additional terms, namely enriched terms, are introduced into the approximated displacement formulation to mathematically describe the discontinuities. Usually, the enriched functions are proposed based on knowledge of closed form asymptotic fields at crack tip, see [4] for isotropic material and [5] for orthotropic materials. Although the enriched functions proposed by [5] for orthotropic materials have been later employed by many authors [6, 7, 8, 9], they are not sufficient in the special case of isotropic materials. In the last few years, a new type of enriched functions, namely ramp functions, was proposed [10], which is not based on analytical solution. However, the ramp-type function is currently not suitable for orthotropic media as there is no information on material orientation incorporated.

Despite of popularity, XFEM still contains the inherent drawbacks of FEM. For example, the gradient fields calculated by FEM (as well as

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XFEM), e.g. strain and stress, are non-physically discontinuous across nodes. Recently, the consecutive-interpolation procedure (CIP) has been introduced as an improvement for FEM [11, 12, 13], such that the conventional FEM formulation is enhanced by averaged nodal gradient. Desirable properties of CIP include the smooth stress fields and the higher accuracy of field variables due to refined interpolation.

In this paper, the CIP is combined with the enriched functions to model behavior of two-dimensional cracked solids. A slight modification is proposed for the enriched functions originally developed by [5], as an attempt to clear the gap between orthotropic materials and isotropic materials.

The outline of the paper is as follows. A brief on CIP formulation on a particular case of 4-node quadrilateral element is reported in Section 2. Section 3 presents the application of CIP in linear elastic fracture mechanics with the aid of enriched functions. Several numerical examples are investigated in Section 4, in order to demonstrate the accuracy of the proposed approach. Conclusions and remarks are given in Section 5.

2 BRIEF ON FORMULATION OF THE CONSECUTIVE - INTERPOLATION 4-NODE QUADRILATERAL ELEMENT (CQ4)

Details on the formulation of the CQ4 element was previously described by the authors [12]. In this paper, the consecutive-interpolation procedure (CIP) is briefly presented for the sake of completeness. Consider a 2D body in the domain Ω bounded by Γ , which is discretized into non-overlapping sub-domains Ω_e , namely finite elements. Any function $u(\mathbf{x})$ defined in Ω can then be approximated using the CIP as

$$u(\mathbf{x}) \approx \tilde{u}(\mathbf{x}) = \sum_{I=1}^n R_I(\mathbf{x}) \hat{u}_I = \mathbf{R} \hat{\mathbf{u}} \quad (1)$$

where n is the number of nodes $\hat{u}_I$ is the nodal value of function $u(\mathbf{x})$ and R_I is the consecutive-interpolation shape function associated with node I (global index). The vector of shape functions $\mathbf{R}$ is determined by [8]

$$\mathbf{R} = \sum_{I=1}^n (\phi_I \mathbf{N}^{[I]} + \phi_{Ix} \bar{\mathbf{N}}_{,x}^{[I]} + \phi_{Iy} \bar{\mathbf{N}}_{,y}^{[I]}) \quad (2)$$

in which $\mathbf{N}^{[I]}$ is the vector of Lagrange shape functions evaluated at node I ; and $\bar{\mathbf{N}}_{,x}^{[I]}$, $\bar{\mathbf{N}}_{,y}^{[I]}$ are the averaged derivative of Lagrange shape functions with respect to x - and y - directions, respectively. $\bar{\mathbf{N}}_{,x}^{[I]}$ is calculated by (and analogously for $\bar{\mathbf{N}}_{,y}^{[I]}$)

$$\bar{\mathbf{N}}_{,x}^{[I]} = \sum_{e \in S_I} (w_e \cdot \mathbf{N}_{,x}^{[I][e]}), \quad (3)$$

with $\mathbf{N}_{,x}^{[I][e]}$ being the derivative of $\mathbf{N}^{[I]}$ computed in element e , while w_e is a weight function defined by

$$w_e = \frac{\Delta_e}{\sum_{\bar{e} \in S_I} \Delta_{\bar{e}}}, \quad e \in S_I. \quad (4)$$

Here, S_I is the the set of elements interconnected at node i , and Δ_e being the area of element e .

It is important to highlight that the auxiliary functions ϕ_I , ϕ_{Ix} , ϕ_{Iy} have to be developed for each type of elements [11, 12], which is actually not a trivial task. Fortunately, a general formulation to determine auxiliary functions for a wide range of finite elements from one to three dimensions has been recently proposed by [9], resolving the bottleneck. For the sake of completeness, the general formulation of auxiliary functions is shown in the followings.

Given an element e with ne number of nodes, the auxiliary functions associated with the local i^{th} node ($i = 1, 2, 3, \dots, ne$) is calculated by [13]

$$\phi_i = N_i + N_i^2 (\Sigma_1 - N_i) - N_i (\Sigma_2 - N_i^2), \quad (5)$$

$$\phi_{ix} = \sum_{j=1, j \neq i}^{ne} (x_j - x_i) \left(N_i^2 N_j + \frac{1}{2} N_i N_j (\Sigma_1 - N_i - N_j) \right) \quad (6)$$

, where N is the Lagrange shape functions and Σ_1 and Σ_2 are determined by

$$\Sigma_1 = \sum_{i=1}^{ne} N_i, \quad (7)$$

$$\Sigma_2 = \sum_{i=1}^{ne} N_i^2, \quad (8)$$

Replacing x -coordinate by y -coordinate in (8), the function ϕ_{iy} is obtained.

Fig. 1 illustrates the application of CIP approach into Q4 element described particularly in an irregular finite element mesh. The point of interest

$\mathbf{x}$ is located inside a 4-node quadrilateral element, where the four local nodes are subsequently denoted as i, j, k, m . The four sets S_i, S_j, S_k, S_m are established by collecting the elements share the node i, j, k, m , respectively. Once the sets S_i, S_j, S_k, S_m are determined, the consecutive-interpolation shape functions can be calculated through (2). As shown in Fig. 1, the set of nodes that support a point of interest $\mathbf{x}$ used in CIP is in any cases larger than that of the conventional FEM, because it includes not only the nodes of the element containing the point $\mathbf{x}$ but also the nodes of the adjacent elements.

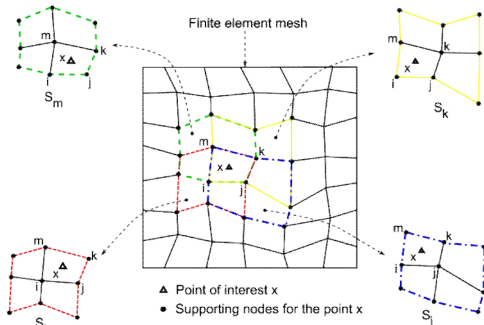


Figure 1. Schematic sketch of CQ4 element

3 APPLICATION OF CIP IN LINEAR ELASTIC FRACTURE PROBLEMS

3.1 Governing equations

The governing equation for static equilibrium in a domain Ω bounded by Γ , assuming small displacements and small strains, is given by

$$\nabla \cdot \boldsymbol{\sigma} + \mathbf{b} = 0, \quad (9)$$

where $\mathbf{b}$ is the body force and $\boldsymbol{\sigma}$ denotes the Cauchy stress tensor. The stress-strain relation is given by Hook's law:

$$\boldsymbol{\sigma} = \mathbf{C} : \boldsymbol{\varepsilon}, \quad (10)$$

in which $\mathbf{C}$ is the fourth-order tensor of material property and the strain tensor $\boldsymbol{\varepsilon}$ is determined by

$$\boldsymbol{\varepsilon} = \frac{1}{2}(\nabla \mathbf{u} + \nabla^T \mathbf{u}), \quad (11)$$

The associated boundary conditions are as follows

$$\mathbf{u} = \bar{\mathbf{u}} \text{ on } \Gamma_u: \text{prescribed displacement}, \quad (12)$$

$$\boldsymbol{\sigma} \cdot \mathbf{n} = \bar{\mathbf{t}} \text{ on } \Gamma_t: \text{prescribed traction}, \quad (13)$$

with $\Gamma = \Gamma_u + \Gamma_t$ and $\Gamma_u \cap \Gamma_t = \{\emptyset\}$. A crack existing in Ω is denoted by Γ_c , which is assumed to be traction free.

3.2 Enriched formulations

In order to mathematically describe the discontinuity, enriched approach [4] is usually

employed. Recently, [14] introduced the extended consecutive-interpolation 4-node quadrilateral element (XCQ4), which incorporates the enriched formulation into CQ4 element, such that the approximated displacement field in Eq. (1) is rewritten by

$$\begin{aligned} \mathbf{u}^h(\mathbf{x}) = & \sum_{i \in I^s} R_i(\mathbf{x}) \hat{\mathbf{u}}_i + \sum_{j \in J^{\text{split}}} R_j(\mathbf{x}) (H(\mathbf{x}) - H(\mathbf{x}_j)) \mathbf{a}_j \\ & + \sum_{k \in K^{\text{tip}}} R_k(\mathbf{x}) \sum_{\alpha=1}^4 (F^\alpha(\mathbf{x}) - F^\alpha(\mathbf{x}_k)) \mathbf{b}_k. \end{aligned} \quad (14)$$

In (14), J^{split} is the set of nodes belong to elements completely cut by crack and K^{tip} is the set of nodes belong to the elements containing the crack tips. The employment of enriched terms lead to additional DOFs $\mathbf{a}_j$ and $\mathbf{b}_k$. Function $H(\mathbf{x})$ is the Heaviside step function, describing the jump in displacement field across the crack, while the four branch functions F^α ($\alpha=1, \dots, 4$) are crack-tip enrichments, capturing the singularities of asymptotic stress fields. For 2D linear isotropic elasticity problems, the four branch functions are given in the local polar coordinate (r, θ) defined at the crack tip by

$$\begin{aligned} F^1 &= \sqrt{r} \sin\left(\frac{\theta}{2}\right), & F^2 &= \sqrt{r} \cos\left(\frac{\theta}{2}\right) \\ F^3 &= \sqrt{r} \sin\left(\frac{\theta}{2}\right) \sin(\theta), & F^4 &= \sqrt{r} \cos\left(\frac{\theta}{2}\right) \sin(\theta) \end{aligned} \quad (15)$$

For 2D linear orthotropic materials, the crack-tip enrichments are introduced as in [5, 11]

$$\begin{aligned} F^1 &= \sqrt{r} \sin\left(\frac{\theta_1}{2}\right) \sqrt{g_1(\theta)}, \\ F^2 &= \sqrt{r} \cos\left(\frac{\theta_1}{2}\right) \sqrt{g_1(\theta)}, \\ F^3 &= \sqrt{r} \sin\left(\frac{\theta_2}{2}\right) \sqrt{g_2(\theta)}, \\ F^4 &= \sqrt{r} \cos\left(\frac{\theta_2}{2}\right) \sqrt{g_2(\theta)}. \end{aligned} \quad (16)$$

Functions $g_q(\theta)$ and θ_q ($q=1,2$) are defined as

$$g_q(\theta) = \sqrt{(\cos \theta + s_{qx} \sin \theta)^2 + (s_{qy} \sin \theta)^2}, \quad (17)$$

$$\theta_q(\theta) = \arctan\left(\frac{s_{qy} \sin \theta}{\cos \theta + s_{qx} \sin \theta}\right), \quad (18)$$

in which $s_q = s_{qx} + is_{qy}$ are the roots of the following characteristic equation

$$C_{11}s^4 - 2C_{13}s^3 + (2C_{12} + C_{33})s^2 - 2C_{33}s + C_{22} = 0 \quad (19)$$

where C_{ij} is the components of the tensor of material property as defined in (10).

When the material is isotropic, i.e. $s_{1x} = s_{2x} = 0$ and $s_{1y} = s_{2y} = 1$, the branch functions calculated using (16,17,18) degenerate into

$$F^1 = F^3 = \sqrt{r} \sin\left(\frac{\theta}{2}\right) \quad \text{and}$$

$$F^2 = F^4 = \sqrt{r} \cos\left(\frac{\theta}{2}\right), \text{ which are clearly not}$$

sufficient as a set of basis functions. Thus, a modification for functions g_I and g_2 in (17) is proposed as follows

$$g_1(\theta) = \sqrt{(\cos\theta + s_{1x}\sin\theta)^2 + (s_{1y}\sin\theta)^2} \quad (20)$$

$$g_2(\theta) = \sqrt{0.5(\sin\theta + s_{2x}\cos\theta)^2 + 0.5(s_{2y}\sin\theta)^2}$$

With equation (20), the enriched functions degenerates exactly into (15) in the special case of isotropic material.

3.3 Computation of Stress Intensity Factors

Stress Intensity Factors (SIFs) are important parameters reflecting the singular fields near the crack tip in linear elastic fracture mechanics. Numerically, SIFs can be determined by the following relation:

$$M^{(1,2)} = 2d_{11}K_I^{(1)}K_I^{(2)} + d_{12}(K_I^{(1)}K_{II}^{(2)} + K_{II}^{(1)}K_I^{(2)}) + 2d_{22}K_{II}^{(1)}K_{II}^{(2)}, \quad (21)$$

in which K_I , K_{II} are the mode I and mode II SIFs, respectively. Subscript (I) denotes the present state of the cracked body, while subscript (2) denotes an auxiliary state, which can be chosen as the asymptotic fields of pure mode I (i.e., $K_I^{(2)} = 1$ and $K_{II}^{(2)} = 0$) or pure mode II (i.e., $K_I^{(2)} = 0$ and $K_{II}^{(2)} = 1$), see [4, 6, 7]. Quantities d_{11} , d_{12} and d_{22} are computed by

$$d_{11} = -\frac{C_{22}}{2} \operatorname{Im}\left(\frac{s_1 + s_2}{s_1 + s_2}\right)$$

$$d_{12} = -\frac{C_{22}}{2} \operatorname{Im}\left(\frac{1}{s_1 s_2}\right) + \frac{C_{11}}{2} \operatorname{Im}(s_1 s_2), \quad (22)$$

$$d_{22} = \frac{C_{11}}{2} \operatorname{Im}(s_1 + s_2)$$

$M^{(1,2)}$ is a path-independent integral, namely interactive integral calculated as follows [6, 7]

$$M^{(1,2)} = \int_{\Gamma} \frac{1}{2} \left(\sigma_{ij}^{(1)} \varepsilon_{ij}^{(2)} + \sigma_{ij}^{(2)} \varepsilon_{ij}^{(1)} \right) d\Gamma$$

$$- \int_{\Gamma} \left(\sigma_{ij}^{(2)} \frac{\partial u_i^{(1)}}{\partial x_1} + \sigma_{ij}^{(1)} \frac{\partial u_i^{(2)}}{\partial x_1} \right) n_j d\Gamma \quad (23)$$

Here, Γ is an arbitrary contour surround the crack tip, which encloses no other types of discontinuities, and n_j is the j^{th} component of the outward unit vector normal to Γ .

4 NUMERICAL EXAMPLES

Four isotropic and orthotropic problems are examined in this section to assess the accuracy and performance of the proposed method. All the numerical examples are listed in the following for clarity:

- Finite rectangular isotropic plate with an edge crack
- Finite rectangular orthotropic plate with an edge crack
- Finite rectangular orthotropic plate with a central slanted crack
- An inclined center crack in an orthotropic disk subjected to point loads

The standard extended 4-node quadrilateral element is denoted by XQ4 and XCQ4 denotes the extended consecutive-interpolation 4-node quadrilateral element. Note that the modified enriched functions in (20) are used by default, otherwise, it is stated clearly whether equation (20) or (17) are used.

4.1 Finite isotropic rectangular plate with an edge crack

An isotropic plate with an edge crack under uniform tensile loading $\sigma_0 = 1$ is considered in this problem, see Fig. 2. The purpose of this example is to demonstrate that the modified branch functions using (20) performs better than the ones originally proposed by Asadpoure and Mohammadi [5] in case of isotropic material. The plate is determined

by $L = 2W = 16$ and a crack length a . Material parameters are given by: Young's modulus $E = 1000$ and Poisson's ratio $\nu = 0.3$. This problem is pure mode I, in which the closed form stress intensity factor K_I is given by [4]

$$K_I = C\sigma_0\sqrt{\pi a}, \quad (24)$$

where

$$C = 1.12 - 0.231\bar{a} + 10.55\bar{a}^2 - 21.72\bar{a}^3 + 30.79\bar{a}^4, \quad (25)$$

$$\bar{a} = \frac{a}{L}. \quad (26)$$

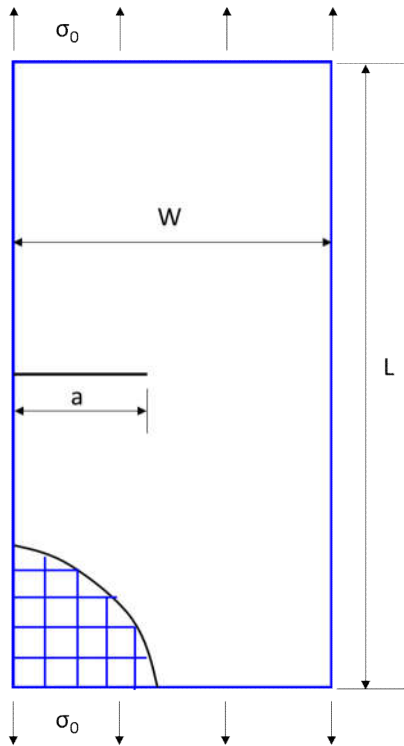


Figure 2. Example 4.1 Isotropic rectangular plate with an edge crack under uniform tensile loading

For numerical calculation, a mesh of 25×49 quadrilateral elements (1300 nodes) is used to discretize the problem domain. The values of K_I evaluated for different ratios a/W are presented in Table 1, in which a comparison between XQ4-(17), XQ4-(20), XCQ4-(20). Analytical solution is used as reference to assess the accuracy of three approaches. Results indicate that the evaluation of K_I by XQ4-(20) is closer than XQ4-(17), evidently showing the appropriateness of the modified enriched function in (20). The highest accuracy in Table 1 is achieved by XCQ4-(20), demonstrating

that XCQ4 element, with the enhanced consecutive-interpolation, outperforms the XQ4 element. Thanks to the consecutive-interpolation procedure, the stress fields evaluated by XCQ4 elements are smooth across element nodes (except for the regions containing crack), which is physically more appropriate than the non-smooth stress provided by XQ4 elements, as depicted in Fig. 3 for the normal stress component σ_{yy} . This is the reason that higher accuracy for SIF values, which are based on stress components, is obtained when XCQ4 elements are used.

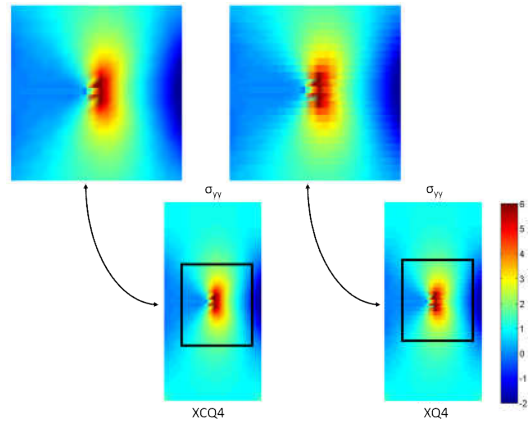


Figure 3. Example 4.1 Normal stress fields σ_{yy} obtained by XCQ4 elements (left) and XQ4 elements (right)

TABLE 1. EXAMPLE 4.1: VALUES OF K_I CALCULATED WITH DIFFERENT CRACK LENGTHS

a/W	Exact	XQ4-(15)	XQ4-(20)	XCQ4 - (20)
0.3	4.558	4.441	4.461	4.467
0.4	6.669	6.522	6.548	6.563
0.5	10.019	9.651	9.733	9.767

4.2 Finite orthotropic rectangular plate with an edge crack

A rectangular orthotropic plate with an edge crack subjected to distributed load, as shown in Fig. 4, is investigated in this problem. The material is made from graphic-epoxy with the following properties: $E_1 = 114.8$ GPa, $E_2 = 11.7$ GPa, $G_{12} = 9.66$ GPa, $\nu_{12} = 0.21$. The crack length is determined by $a/W = 0.5$. The same mesh of 25×49 quadrilateral elements as in Example 4.1 is used for discretization of the problem domain.

Effects of the material orthotropic angle β on mixed-mode stress intensity factors are shown in Fig. 5. The present approach is in good agreement with reference results [5, 6, 15]. K_I tends to increase from $\beta = 0^\circ$ to $\beta = 45^\circ$ and decreases from $\beta = 90^\circ$.

For K_{II} , the peak value is reached at about $\beta = 30^\circ$. In Fig. 5, the SIFs are normalized by

$$\tilde{K}_I = \frac{K_I}{\sigma_0 \sqrt{\pi a}}, \quad (27)$$

$$\tilde{K}_{II} = \frac{K_{II}}{\sigma_0 \sqrt{\pi a}}, \quad (28)$$

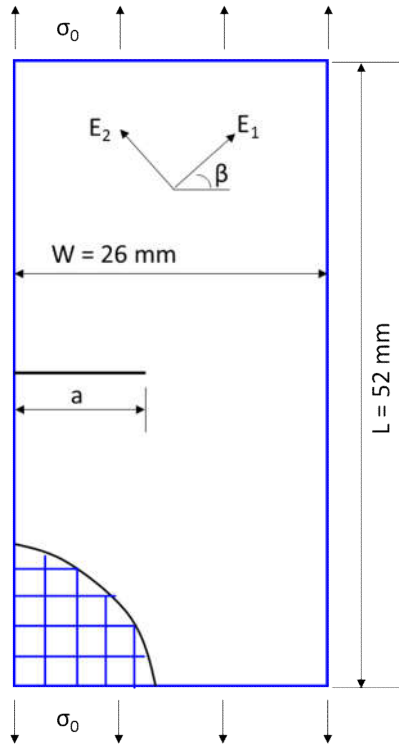


Figure 4. Example 4.2: Orthotropic rectangular plate with an edge crack under uniform tensile loading

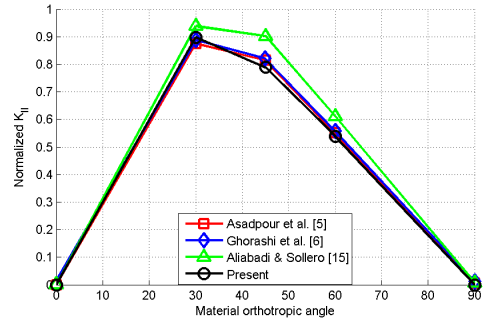
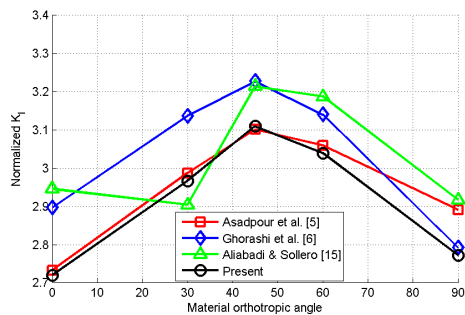


Figure 5. Example 4.2: Normalized mode I and mode II SIFs computed according to the material orthotropic angle

4.3 Finite orthotropic rectangular plate with a slanted center crack

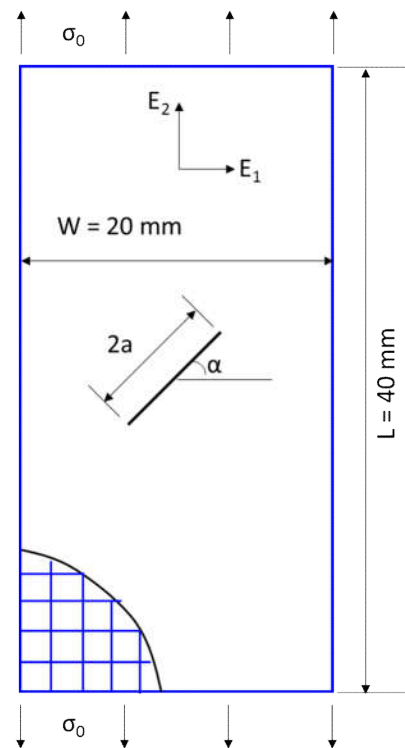


Figure 6. Example 4.3: Orthotropic rectangular plate with a slanted center crack

In this example, the mixed-mode problem of a finite rectangular plate with a slanted center crack is investigated, see Fig. 6. The geometry is given by $L = 2W = 40$ and the crack length is $2a = 2\sqrt{2}$. The orthotropic material axes are aligned with the global coordinate x- and y- axes. The orthotropic material parameters are as follows: $E_1 = 35$ GPa, $E_2 = 12$ GPa, $G_{12} = 3$ GPa, $\nu_{21} = 0.07$.

The problem domain is discretized by a mesh of 45×91 quadrilateral elements (i.e. 4232 nodes). Fig. 7 depicts the variation of normalized mode I and mode II SIFs with respect to the inclined angle of crack. As the inclined angle increased from 0° to 90° , mode I SIF, $\tilde{K}_I$, gradually decreases from 1 to 0, while the mode II SIF, $\tilde{K}_{II}$, increases to the peak value at 45° and then decreases to 0. Good agreement with results using meshfree method (4560 nodes) reported in [6] is observed. Largest discrepancy in Fig. 7 is recorded between the curves of $\tilde{K}_{II}$ at slanted angle $\alpha = 45^\circ$. Thus, further comparison is conducted and reported in Table 2, showing the consistency between present approach (XCQ4 - (20)) and literatures.

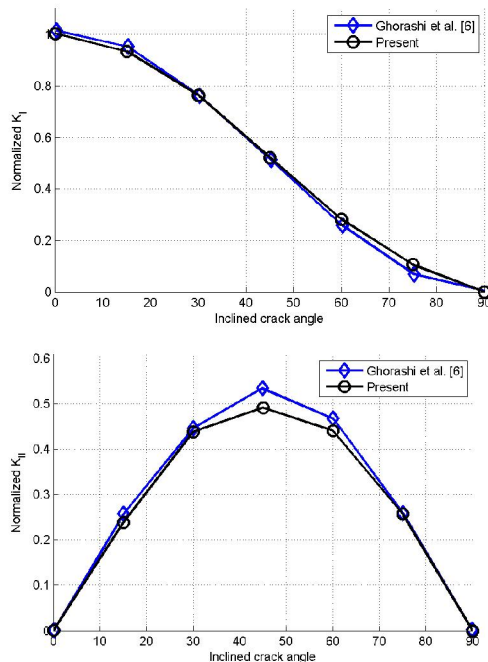


Figure 7. Example 4.3: Normalized mode I and mode II SIFs computed according to the crack inclined angle

TABLE 2. EXAMPLE 4.3: COMPARISON OF NORMALIZED MODE I AND MODE II SIFS AT INCLINED CRACK ANGLE 45°

	Present	[6]	[16]	[17]
$\tilde{K}_I$	0.522	0.512	0.485	0.5
$\tilde{K}_{II}$	0.491	0.530	0.498	0.5

5 CONCLUSIONS AND OUTLOOKS

In this paper, the XCQ4 element has been successfully extended for modelling cracks in two-dimensional orthotropic problems. The accuracy and performance of the present formulation has been verified through a series of numerical examples. Preliminary results indicate that the present approach is in good agreement with other

authors. Furthermore, XCQ4 element is observed to perform better than its XFEM counterpart, the XQ4 element, such that higher accuracy of SIFs is achieved. As SIFs are key quantities to numerically determine the propagating direction during crack advancement, the approach is promising to be extended to problems involving crack growth.

The higher accuracy of XCQ4 over XQ4 is possibly due to the enhanced interpolation by CIP, by which the erroneous non-smooth stress fields in XQ4 can be overcome by XCQ4. It is important to emphasize that no extra degrees of freedom is required for CIP. Although in this work, only the quadrilateral element is investigated, the approach is possible for other types of element. With the aid of general formulation for auxiliary functions, see [13], CIP can be integrated into a wide range of existing finite elements without difficulties.

The set of crack-tip enriched functions proposed by [5] is shown to be not well-chosen. Thus, a modified version of the enriched functions is presented, which properly degenerates into those proposed by [4] for the special case of isotropic material. The new set of enriched functions outperforms the set by [5] when material is isotropic. For orthotropic material, the new set of enriched functions is consistent with references available in literatures.

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Một phương pháp số mới cho bài toán vết nứt trong vật liệu trục hướng

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Tóm tắt — Bài báo trình bày một phương pháp số mới cho bài toán phân tích vết nứt trong miền hai chiều với vật liệu trục hướng. Phương pháp được đề xuất dựa trên kỹ thuật nội suy liên tiếp và hàm làm giàu. Kỹ thuật nội suy liên tiếp là kỹ thuật mới, được giới thiệu trong vài năm gần đây để cải tiến phương pháp phần tử hữu hạn. Theo đó, lời giải thu được có độ chính xác và độ liên tục bậc cao hơn mà không làm tăng số bậc tự do. Khi áp dụng cho bài toán vết nứt, để tránh việc chia lưới lại, kỹ thuật hàm làm giàu được áp dụng để mô tả bước nhảy trong miền chuyển vị và sự suy biến ứng suất quanh đỉnh vết nứt bằng hàm toán học.

Độ chính xác của phương pháp khi phân tích vết nứt trong miền hai chiều với vật liệu trục hướng sẽ được khảo sát qua các ví dụ tính toán khác nhau. Giá trị hệ số cường độ ứng suất sẽ được so sánh kiểm chứng với các lời giải tham khảo.

Từ khóa — Kỹ thuật nội suy liên tiếp, phân tích vết nứt, hàm làm giàu, vật liệu trục hướng, hệ số cường độ ứng suất.